

1 1D waves

1.1 Wave propagation

$$\frac{\partial u^2}{\partial t^2} = c^2 \frac{\partial u^2}{\partial x^2}$$

$$u(x, t) = f(x + ct) + g(x - ct)$$

$$\epsilon = \frac{\partial u}{\partial x}$$

$$v = \frac{\partial u}{\partial t}$$

1.2 Impedance

$$R = \frac{\rho_2 c_2}{\rho_1 c_1}$$

$$f_T = \frac{2R}{R+1} f_I$$

$$f_R = \frac{R-1}{R+1} f_I$$

$$f_I + f_R = f_T$$

1.3 Energy

$$E_{pot} = \frac{\sigma}{2} \int_0^L \epsilon dx$$

$$E_{kin} = \frac{\rho}{2} \int_0^L v^2 dx$$

1.4 Harmonic waves

$$u(x, t) = A \cos(kx \pm \omega t) = A e^{i(\omega t - kx)}$$

$$\omega = \frac{2\pi}{T} = 2\pi f$$

$$\lambda = cT$$

$$k = \frac{2\pi}{\lambda} = \frac{\omega}{c}$$

$$c = \frac{\omega}{k} = \frac{\lambda}{T}$$

2 3D waves

2.1 Wave propagation

$$[(\lambda + \mu) \underline{p} \otimes \underline{p} + \mu (\underline{p} \cdot \underline{p}) \underline{I}] \underline{d} = \rho c^2 \underline{d}$$

$$\underline{d} = d \underline{p} \rightarrow C_L = \sqrt{\frac{\lambda + 2\mu}{\rho}} = \sqrt{\frac{E(1-\nu)}{\rho(1+\nu)(1-2\nu)}}$$

$$d \cdot \underline{p} = 0 \rightarrow C_T = \sqrt{\frac{\mu}{\rho}} = \sqrt{\frac{E}{2\rho(1+\nu)}}$$

2.2 Rayleigh waves

$$u_1 = A e^{-bx_2} e^{ik(x_1 - ct)}$$

$$u_2 = B e^{-bx_2} e^{ik(x_1 - ct)}$$

$$u_3 = 0$$

2.3 Helmholtz decomposition

$$\underline{u} = \underline{u}_L + \underline{u}_C = \nabla \varphi + \text{rot} \underline{\psi}$$

3 Linear Elastic Fracture Mechanics

3.1 Griffith's energy criterion (Global)

$$\frac{dE}{dA} = \frac{dE_{pot}}{dA} + \frac{dW}{dA}$$

$$\sigma_f = \sqrt{\frac{2E\gamma_s}{\pi a}}$$

$$a_{Griffith} = \frac{2E\gamma_s}{\pi \sigma^2}$$

3.2 Energy release rate

$$G = -\frac{dE_{pot}}{dA} = \frac{1}{B} \frac{dU}{da} = \frac{1}{B} \frac{d}{da} (W - U_{el})$$

$$C = \frac{\Delta}{P}$$

$$G = \frac{P^2}{2B} \frac{dC}{da}$$

3.3 Airy function

$$\nabla^4 \phi = 0$$

$$\sigma_{xx} = \frac{\partial^2 \phi}{\partial y^2}$$

$$\sigma_{yy} = \frac{\partial^2 \phi}{\partial x^2}$$

$$\sigma_{xy} = -\frac{\partial^2 \phi}{\partial x \partial y}$$

3.4 Irwin's solution (Local)

3.4.1 Mode I

$$\sigma_{ij}(r, \theta) \simeq \frac{K_I}{\sqrt{2\pi r}} f_{ij}(\theta)$$

$$f_{xx} = \cos\left(\frac{\theta}{2}\right) \left[1 - \sin\left(\frac{\theta}{2}\right) \sin\left(\frac{3\theta}{2}\right)\right]$$

$$f_{yy} = \cos\left(\frac{\theta}{2}\right) \left[1 + \sin\left(\frac{\theta}{2}\right) \sin\left(\frac{3\theta}{2}\right)\right]$$

$$f_{xy} = \cos\left(\frac{\theta}{2}\right) \sin\left(\frac{\theta}{2}\right) \cos\left(\frac{3\theta}{2}\right)$$

$$f_{zz}(\text{Plane strain}) = \nu(f_{xx} + f_{yy})$$

$$f_{zz}(\text{Plane stress}) = f_{xz} = f_{yz} = 0$$

3.4.2 Mode II

$$\sigma_{ij}(r, \theta) \simeq \frac{K_{II}}{\sqrt{2\pi r}} f_{ij}(\theta)$$

$$f_{xx} = -\sin\left(\frac{\theta}{2}\right) \left[2 + \cos\left(\frac{\theta}{2}\right) \cos\left(\frac{3\theta}{2}\right)\right]$$

$$f_{yy} = \sin\left(\frac{\theta}{2}\right) \cos\left(\frac{\theta}{2}\right) \cos\left(\frac{3\theta}{2}\right)$$

$$f_{xy} = \cos\left(\frac{\theta}{2}\right) \left[1 - \sin\left(\frac{\theta}{2}\right) \sin\left(\frac{3\theta}{2}\right)\right]$$

$$f_{zz}(\text{Plane strain}) = \nu(f_{xx} + f_{yy})$$

$$f_{zz}(\text{Plane stress}) = f_{xz} = f_{yz} = 0$$

3.5 K values

Crack, length $2a$, in an infinite plate:

$$K_I = \sigma \sqrt{\pi a}$$

Same shape, with a symmetric stress distribution $\sigma(x)$ along y:

$$K = 2\sqrt{\frac{a}{\pi}} \int_0^a \frac{\sigma(x)}{\sqrt{a^2 - x^2}} dx$$

Crack, length a , on the edge of a semi-infinite plate:

$$K_I = 1.12 \sigma \sqrt{\pi a}$$

Crack of length $2a$ in an infinite plate, with an angle β :

$$K_I = \sigma \cos^2 \beta \sqrt{\pi a}$$

$$K_{II} = \sigma \sin \beta \cos \beta \sqrt{\pi a}$$

Central penny-shaped crack, radius a , in an infinite body:

$$K_I = \frac{2}{\pi} \sigma \sqrt{\pi a} = 2\sigma \sqrt{\frac{a}{\pi}}$$

Same shape, with a symmetric stress distribution $\sigma(r)$ along y:

$$K = \frac{2}{\sqrt{\pi a}} \int_0^a \frac{r \sigma(r)}{\sqrt{a^2 - r^2}} dr$$

Decomposition of a problem into subparts: $K_I^{tot} = K_I^A + K_I^B + K_I^C \dots$

3.6 K-G relation

$$G = \frac{K_I^2}{E'} + \frac{K_{II}^2}{E'} + \frac{K_{III}^2}{2\mu}$$

$$\text{Plane strain : } E' = \frac{E}{1-\nu^2}$$

$$\text{Plane stress : } E' = E$$

3.7 Mixed-Mode Fracture

Principle of Local Symmetry:

$$K_{III}^*(\alpha^*) = 0$$

Maximum Hoop Stress:

$$\sigma_{\theta\theta}(\alpha) \leq \sigma_{\theta\theta}(\alpha^*), \forall \alpha$$

Maximum Energy Release Rate Criterion:

$$G(\alpha) \leq G(\alpha^*), \forall \alpha$$

4 Brittle materials

$$S(V, \sigma) = \prod_{i=1}^N S(\Delta V_i, \sigma)$$

2-parameters Weibull distribution:

$$F(V, \sigma) = 1 - S(V, \sigma) = 1 - \exp \left[- \int_V \left(\frac{\sigma(V)}{\sigma_0} \right)^m \frac{dV}{V_0} \right] = 1 - \exp \left[-k_m \frac{V}{V_0} \left(\frac{\sigma_{max}}{\sigma_0} \right)^m \right]$$

Tensile: $k_m = 1$

$$\text{3-point bending: } k_m = \frac{1}{2(m+1)^2}$$

$$\text{4-point bending: } k_m = \frac{mL_i + L_0}{2L_0(m+1)^2}$$

5 Plasticity in Fracture Mechanics

$$\sigma_{VM} = \frac{1}{\sqrt{2}} \left[(\sigma_1 - \sigma_2)^2 + (\sigma_1 - \sigma_3)^2 + (\sigma_2 - \sigma_3)^2 \right]^{0.5}$$

$$K_{IC} = \sqrt{EG} = \sqrt{E(2\gamma_s + G_{pl})}$$

1st order approximation: $\sigma_{yy} = \sigma_{ys}$

$$r_p = \frac{1}{2\pi} \left(\frac{K_I}{\sigma_{ys}} \right)^2$$

2nd order approximation:

$$r_p = \frac{1}{\pi} \left(\frac{K_I}{\sigma_{ys}} \right)^2$$

J-integral (for pulling in the y-direction):

$$J = \int_{\Gamma} (w dy - T_i \frac{\partial u_i}{\partial x} ds)$$

6 Fatigue crack growth

Paris law:

$$\frac{da}{dN} = C \Delta K^m \text{ with } m \text{ between } 2 \text{ and } 4.$$